

RESONANCE VIBRATIONS OF THE NON-IDEAL SYSTEM HAVING THE MISES GIRDER AS ABSORBER

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ABSTRACT

The systems with a limited power supply are characterized by interaction of source of energy and elastic sub-system which is under action of the source. Such systems are named also as the non-ideal systems (NIS). The most important effect observed in the NIS is the Sommerfeld effect [1], when the large amplitude regime is appeared in the elastic sub-system, and the large part of the vibration energy passes from the energy source to the resonance vibrations. Resonance dynamics of such systems was first analytically described by V.O.Kononenko [2]. Investigations on the subject were continued in different publications. In particular, overviews on numerous studies of the non-ideal systems dynamics can be found in [3,4]. Here a resonance behavior of NIS containing the source of energy, the linear sub-system and the nonlinear absorber (Fig.1), is analyzed by the multiple scales method. The snap-through truss is chosen as absorbers. Two resonance regimes of vibrations are considered, namely, the resonance between the linear sub-system and the motor, and the resonance between the absorber and the motor. The multiple scales method is used to describe the system resonance behavior. Corresponding frequency responses are constructed. The snap-through motions and outcome from the resonance behavior in the system are considered too.

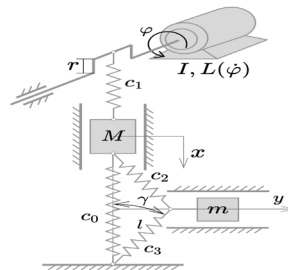


Figure 1: Non-ideal system having Mises girder as absorber

Keywords: Non-ideal system, Mises girder, Resonance vibrations...

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VORTEX-INDUCED VIBRATIONS OF RISERS IN OCEANS CURRENTS

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ABSTRACT

Dynamic behaviour of structures of an increased aspect ratio in a fluid flow depends on the flow velocity profile along the length, and it is important to consider its possible variations. In this work, the model of vortex-induced vibrations of risers [1, 2] previously proposed to describe the structural dynamics in the uniform flow is developed for non-uniform flows. The considered structure, shown in Fig. 1a, has pinned connections and is free to oscillate in both in-line and cross-flow directions, subjected to both linear and nonlinear velocity profiles, as in Fig. 1b. The dimensionless vortex shedding frequency $\Omega_R(\zeta)$ is considered varying with the dimensionless coordinate ζ along the length and constant in time, or:

$$\Omega_R(\zeta) = St(U_R(\zeta) - K), \quad (1)$$

where $U_R(\zeta)$ is the height-dependent profile of the reduced velocity, St stands for the constant Strouhal number, and K is the empirically defined delay coefficient.

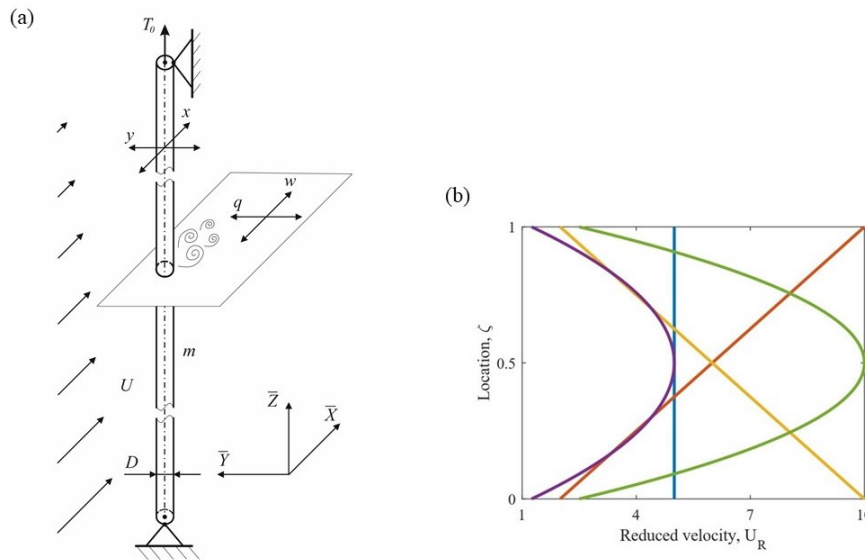


Figure 1: Flexible riser and sheared flow model: a) vortex-induced vibrations of flexible structure in sheared flow; b) reduced velocity variation along the axial coordinate ζ .

The integral of the velocity profile $U_R(\zeta)$ is incorporated into modal interaction coefficients obtained during the Galerkin procedure. The system of equations includes Krenk-Nielsen - van der Pol and van der Pol oscillators in order to model lift and drag force fluctuations coupled with the structural motion. The system is solved with the ode45 MATLAB solver, and the results show changes in the predicted dynamics of a flexible structure with a high aspect ratio related to the flow velocity variations and tension effects.

Keywords: Vortex-induced vibrations, Flexible risers, Pipeline dynamics, Ocean currents, Sheared flow

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SUPPRESSION OF RESONANCE VIBRATIONS OF RECTANGULAR PLATE INTERACTING WITH FLUID

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ABSTRACT

Thin plates have found their application in modern engineering structures, which are in contact with a fluid or completely immersed in it. Often they are important components and operate under conditions of high loads, vibrations or seismic forces. The risk of resonance phenomena generates the need to monitor the current conditions of the object, and to control the occurring dynamic processes. This will prevent emergency situations causing instant or fatigue failure of the structure. For known and time-constant frequency of external excitation or self-exciting undamped vibrations, their effective suppression can be achieved by the application of oppositely directed external harmonic force or using piezoelectric shunt damping technology. The advantage of the last approach is the possibility of executing continuous control of the dynamic behavior of structures, including underwater constructions.

The effect of the added mass and dissipative mechanism arising from the hydroelastic interaction can significantly affect the dynamic processes occurring in the structures. In this work, the influence of the above factors on the efficiency of damping the resonance vibrations of a rectangular plate interacting with quiescent fluid is studied experimentally. The vibrations are excited by an electromagnetic force, which is generated by the interaction of a light neodymium magnet glued to the bottom surface of the plate and a coil located in the neighborhood. To reduce the amplitude of resonant vibrations we used the piezoelectric element glued to the opposite (underwater) surface of the plate and connected to the electrical impedance in the form of external passive electric circuit. The study was carried out in several steps. First, we determined the natural frequencies of the structure vibrations for the operating modes of the piezoelectric element with open-circuited and short-circuited electrodes. Then, the results of numerical simulation [1] or analytical calculations [2] were used to select the values of the resistance and inductance parameters of the external RL -circuit, which provided the most effective damping of harmonic vibrations at a given frequency. The physical coil in the shunt circuit was replaced by a compact inductance equivalent circuit — a gyrator. Its main function is to make the input voltage and current of the circuit behave like the voltage and current in an inductor using for this purpose a capacitor and operational amplifiers. At the final stage, forced harmonic vibrations of the plate were excited. The Polytec PDV-100 laser vibrometer was used to measure the velocity at a reference point on the surface of the structure before and after the piezoelectric element was connected to a shunt circuit.

The experimental studies have shown that the resonant vibrations of the plate interacting with fluid layer cannot be completely suppressed by means of a piezoelectric element connected to an external passive electric circuit. The efficiency of damping decreases more than twice as the height of the fluid layer increases. The variants of attaching different types of piezoelectric elements to the structure surface are analyzed. It has been found that the application of a thin layer of viscous glue eliminates most of the positive effect. As an alternative, we consider the case when a piezoelectric element is connected to an alternating voltage source rather than to a shunt circuit. By selecting its amplitude at a frequency known in advance or determined by the device, it is possible to achieve almost complete suppression of vibrations.

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Keywords: Fluid-structure interaction, Vibrations damping, Plate, Shunted piezoelectric element.

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NUMERICAL BIFURCATION ANALYSIS OF AN OSCILLATING AEROFOIL UNDER NOISE-LOAD

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ABSTRACT

Tracing steady-state solutions in nonlinear experiments or stochastic numerical models is a challenging task as the standard techniques of bifurcation analysis, such as numerical continuation, cannot be directly employed. Furthermore, the random perturbation may result in a loss of information about the underlying deterministic system.

In our study, we employ control-based continuation (CBC) to investigate the dynamics of a numerical model of an aerofoil with pitch and heave degrees-of-freedom [1]. This model is capable to capture the self-excited oscillations associated with aeroelastic flutter and can predict the critical airspeed corresponding to the stability boundary. It can also feature a subcritical Hopf bifurcation, generating a region of airspeeds where two coexisting stable steady-state solutions occur for the same set of parameters: an equilibrium and a large amplitude limit cycle.

Since flutter involves a highly unsteady airflow around the aerofoil, often considerable process noise occurs in experiments. In our numerical model, this is emulated by considering a Langevin-type stochastic moment acting on the aerofoil.

Studying a periodically forced oscillator, Schilder et al. [2] addressed the challenges of tracing periodic solutions in the presence of noise by making their algorithm robust against random perturbations while they perform continuation directly on the noise-polluted experiment. In contrast, our aim is to use a representative surrogate model of the stochastic process and to include parameter identification as part of the continuation algorithm. Thus, we perform nonlinear root finding on a deterministic model, regularly updated to ensure accordance with the modelled process instead of directly using the noise-polluted time-profiles from the simulation of the stochastic system.

As shown in Fig. 1, the presence of noise affects the bifurcation diagram by resulting in a drift from the deterministic equilibria and limit cycles. Having a deterministic surrogate model of the stochastic process can help to capture the deterministic response more accurately while it may also reveal more information about the solutions by determining their stability or identifying bifurcations, which may not be possible to extract robustly based on the noise-polluted time-profiles.

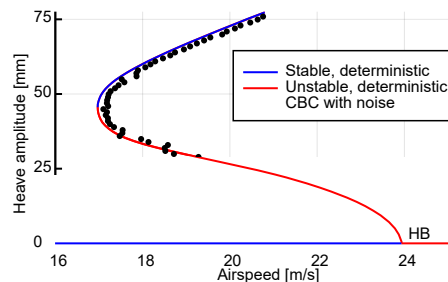


Figure 1: Bifurcation diagram of the deterministic and stochastic models of aeroelastic flutter.

Keywords: control-based continuation, flutter, process noise, surrogate model, bifurcation analysis.

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EFFECT OF DAMPING ON VORTEX-INDUCED OF A CIRCULAR CYLINDER ON THE NONLINEAR SPRING SUPPORT

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ABSTRACT

Vortex-induced vibration (VIV) occurs when shedding vortices (a von Karman vortex street) exert oscillatory forces on a cylinder in the direction perpendicular to both the flow and the structure. Effect of damping on vortex-induced vibration of circular cylinder supported on non-linear spring have been studied numerically for the fixed mass ratio ($m^* = 2.546$) at Reynolds no, $Re = 150$. For the classic case where structure support is made of spring and damper in parallel (fig. 1(i) (a)), this study consider two spring and one damper system as shown in fig. 1(i) (b), motivated by SLS (standard linear solid) model of viscoelasticity [1]. The non linear structural system is governed by the following three parameters: (a) the ratio of springs constant ($R = k_n/k$), (b) damping ratio ($\zeta = c/\sqrt{km}$), and (c) non-linearity strength ($\lambda = -D/\sqrt{-k/a}$) [2]. The VIV response of a circular cylinder in uniform flow is characterized by the Reynolds number ($Re = UD/\nu$), the mass ratio, the damping ratio and the reduced velocity ($U_r = U/f_s D$), where f_s is the natural frequency of nonlinear structural system. The damping effect on VIV response shows the non-monotonic variation with ζ .

Fig. 1(ii) (a) shows amplitude of cylinder displacement, $A_{y,max}^*$ increases continuously upto maximum value and suddenly drop with U_r , the location of peak amplitude non-monotonically vary with ζ . The peak amplitude for $\zeta = 0.001$ occurs at $U_r = 8.4$, decreases to $U_r = 7.2$ for $\zeta = 1$, and further increases to $U_r = 8$ for $\zeta = 10$. Fig. 1(ii) (b) shows the peak lift coefficient, $C_{L,max}$ with U_r . Peak lift coefficient is nearly constant for higher amplitude region and the range of U_r corresponding to peak vary non monotonically with ζ . Fig. 1(ii) (c) shows the the normalised frequency (f_y^*), ratio of vibration frequency to the natural frequency of nonlinear damped system. The range of U_r for $f_y^* \approx 1$, is known as the lock-in region. The results show the maximum amplitude, peak lift coefficient and constant frequency value is non-monotonic with damping, contrary to the classic case of VIV (fig. 1(i) (a)) where these parameters monotonically decreases with damping.

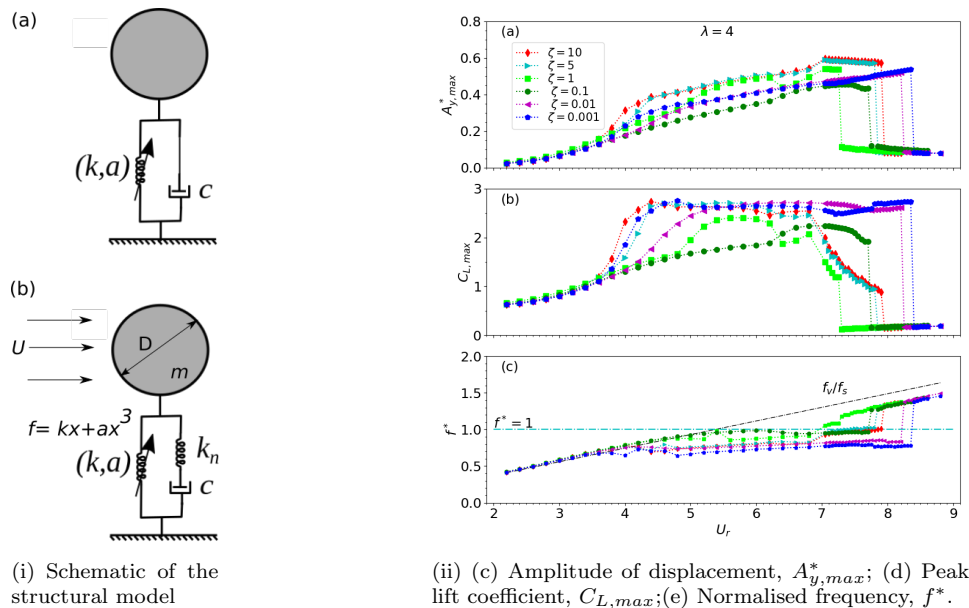


Figure 1: Structural model and VIV response

Keywords: Fluid-Structure Interactions, Vortex-induced vibration, Cubic stiffness non-linearity, Damping

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INVARIANT MANIFOLD FOR THE ANALYSIS OF FLEXIBLE CYLINDERS UNDER VORTEX-INDUCED VIBRATIONS

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ABSTRACT

Vortex-induced vibration (VIV) is a self-induced and self-sustained resonant phenomenon that can be commonly observed in the very slender structures found in the offshore engineering. A review of some topics of research concerning VIV can be found in [1]. From the mathematical point of view, a possible way to model VIV is by coupling a wake-oscillator to the equations of motion of the structure. In the present work, the wake-oscillator model presented in [2] is used in an extended manner, being applied to flexible cylinders. The goal of this paper is to derive invariant manifolds for the system of ordinary differential equations of motion of the problem, obtained from a Galerkin procedure over the equations of motion in the continuum. In [3], the invariant manifold approach is applied to the structural part of the equations of motion only. In [4], VIV of a rigid and elastically mounted cylinder is investigated and a manifold is obtained to represent the wake-oscillator, however being limited to very high structural damping ratios. The present work is an extension of those previous developments. In Fig. 1, the results obtained using the invariant manifold for the wake variable and a direct integration of the system of equations are compared to show the agreement. It is possible to see that the results are in good agreement in the amplitude of response, with a small difference in the oscillation frequency. Further results can then be explored using the invariant manifolds.

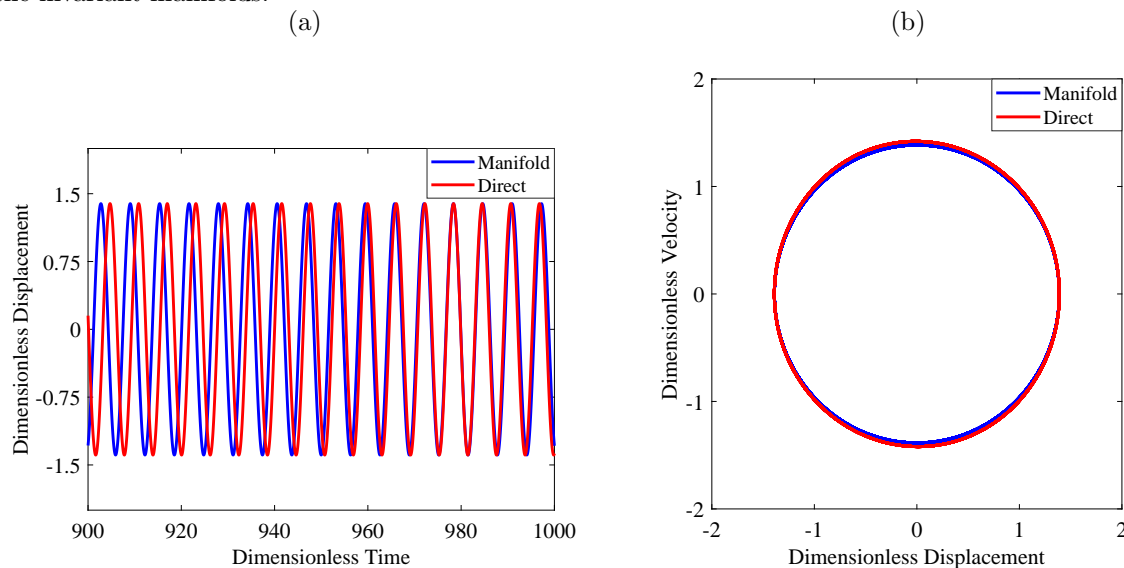


Figure 1: (a) Comparison of the time-series of the transversal displacement of the cylinder. Direct integration of the differential equations and manifold solution; (b) Comparison of the phase space for the transversal displacement and velocity of the cylinder. Direct integration of the differential equations and manifold solution

Keywords: Invariant manifolds, Flexible slender cylinder, Vortex-induced vibration.

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